





REEACH Program Summary

Aviation with low Carbon and Range Extenders for Electric High efficiency

ARPA-E Mission



REDUCE

emissions



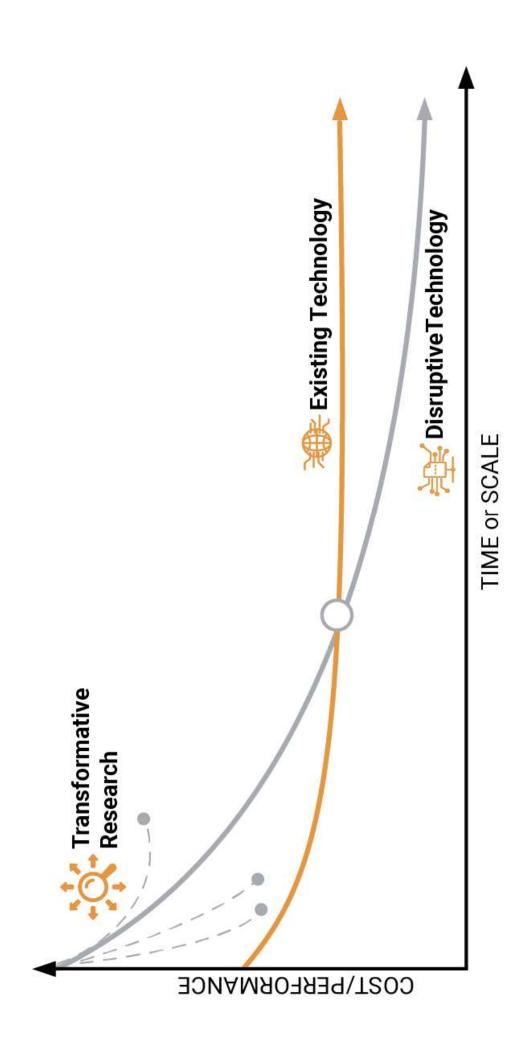






What Problems are We Trying to Solve?





ARPA-E Impact Indicators 2023

ARPA-E has provided Since 2009

\$3.58 billion

more than 1,500 projects in R&D funding to

+ 42 selected projects





in private-sector follow-on funding

149 companies

ARPA-E projects



27 exits

from mergers, acquisitions, and IPOs

300 projects have partered with

for further development other government

agencies

ournal articles peer-reviewed from ARPA-E 6,797 projects



Trademark Office U.S. Patent and patents 1,039 issued by



licenses ARPA-E projects reported from

As of July 2023

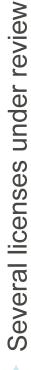
REEACH Indicator additions

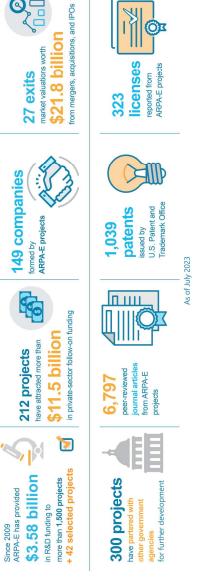


- 42+ patents
- 12+ papers

A

2 startup companies







ARPA-F



James Seaba Viv Program Director Te



a Vivien Lecoustre tor Tech/T2M SETA



Colin Gore Tech SETA

Performers



Mr. Subir Roychoudhury, Precision Combustion, Inc., SOFCS for Flight

Performers



Professor Christopher Cadou, University of Maryland, *Hybrid SOFC-Turbogenerator for Aircraft*



Dr. John Hong,

GE Aerospace Research Center, FueL CelL Embedded ENgine (FLyCLEEN)



Professor Xiao-Dong Zhou,

University of Connecticut,
High Performance Metal-Supported SOFC
System for Range Extension of
Commercial Aviation

∞

Aviation

- Is/will be critical to our economy and quality of life
- Is/would be a significant contributor to fuel consumption— if we don't act
- Passenger-miles-traveled forecast to nearly double between 2016 & 2040 [1]
- Approx. 25% of flying costs is fuel
- Need lightweight economically-attractive climate-friendly propulsion options

Carbon-Neutral Liquid Fuels (CNLF): e.g., Synthetic Aviation Fuel, CH₃0H, NH₃,H₂

High specific-energy batteries

High-efficiency chemical- to thrust- power conversion systems

1. EIA Annual Energy Outlook 2020

Airliner Economics - Cost per Available Seat Mile

Cost per Available Seat Mile (CASM)

airline industry for measuring cost of ► Essential metric in operating an aircraft

Total Operating Expenses

 $\overline{Total Seat Miles Available to Passengers} = Cost per Available Seat Mile$

Primary Factors that affect CASM:

- Fuel prices: Jet A exhibits high price volatility
- Labor costs: Airlines with higher labor costs must cut other services
- Aircraft maintenance costs: Older fleets incur higher maintenance

Reducing CASM

- ▶ Investment: Newer, more fuel-efficient aircraft require less maintenance and fuel
- Fuel Prices: Negotiating better fuel prices and hedging schemes
- Flight Scheduling: Optimizing schedules reduces idle time and aircraft utilization

Airliners' Average Cost per Available Seat Mile¹

(Q1 2024)



Related primarily to ownership of aircraft, ground support equipment, information technology, etc. Depreciation and Amortization 2%

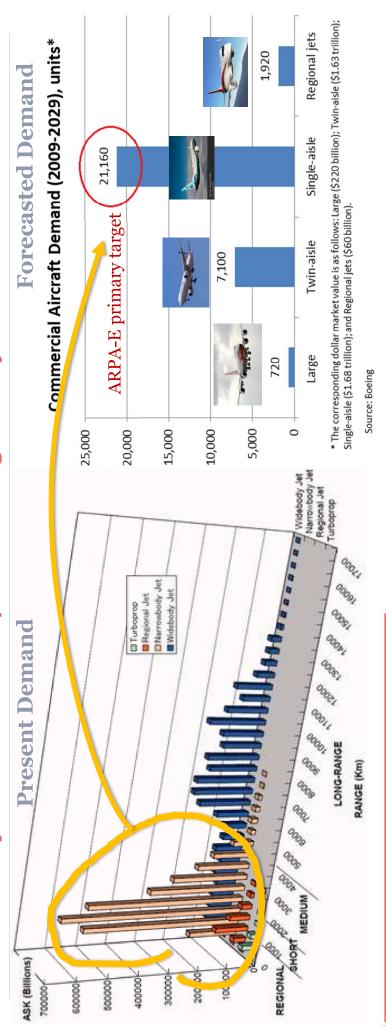
Landing fees and airport (terminal / %9



Maintenance materials and repairs %9 29%

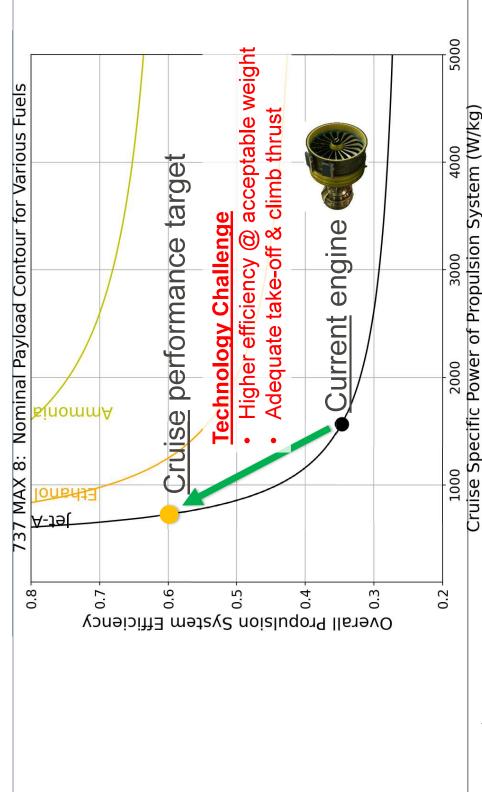


Narrow-body Aircraft Will Keep Dominating the Sky & the Market



Asian demand will be the largest at 6,710 planes, followed by Europe (5,380), North America (5,180), and Latin America (1,800)

Must Achieve High Efficiency at an Acceptable Weight

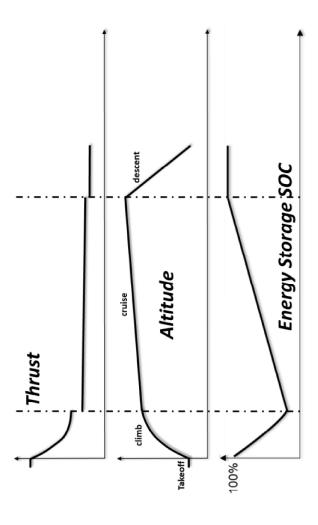


SARPAE

REEACH Program- Metrics & Results

| | E | | | |
|-------------|------------------------|---|--|-----------------------------|
| Target | > 3000 Wh/kg | > 0.75 kW/kg | < \$0.15/kWh | < \$1000/kW |
| Description | System specific energy | Powertrain system specific power > 0.75 kW/kg | Cost of fuel for delivered electric energy | Initial capital system cost |

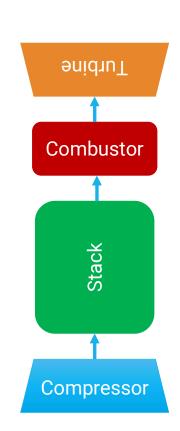
Mission Profile for Modeling



Executive Technical Summary:

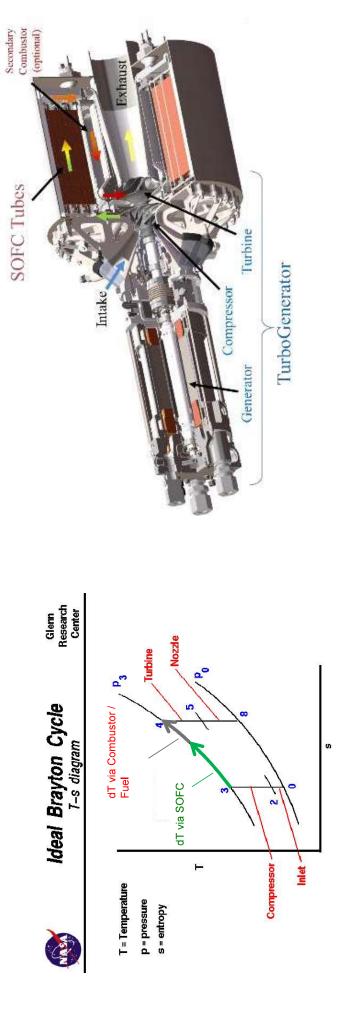
- MS-SOFC is a huge success; met all system targets
- SAF processing integrated successfully with MS-SOFC
- Bottom cycle heat recovery pending

SOFC/Gas Turbine Hybrids

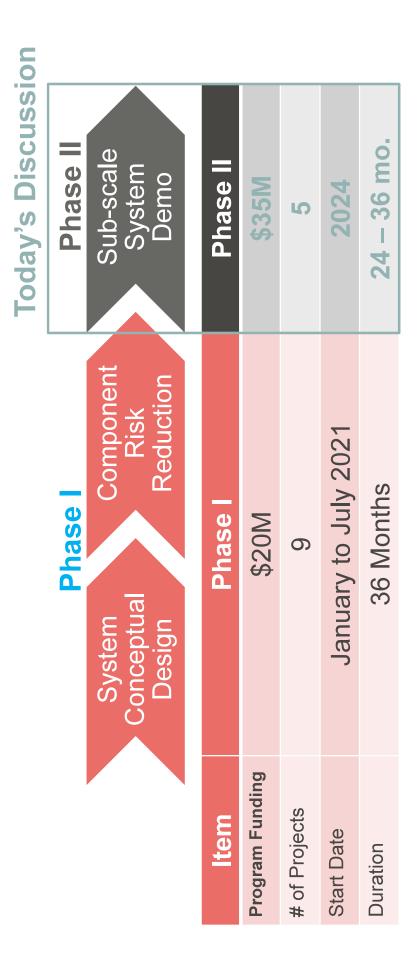


DARPA SHEPARD (Serial Hybrid Electric Propulsion AiRcraft Demonstrator)

SOFC system integration for HEVP

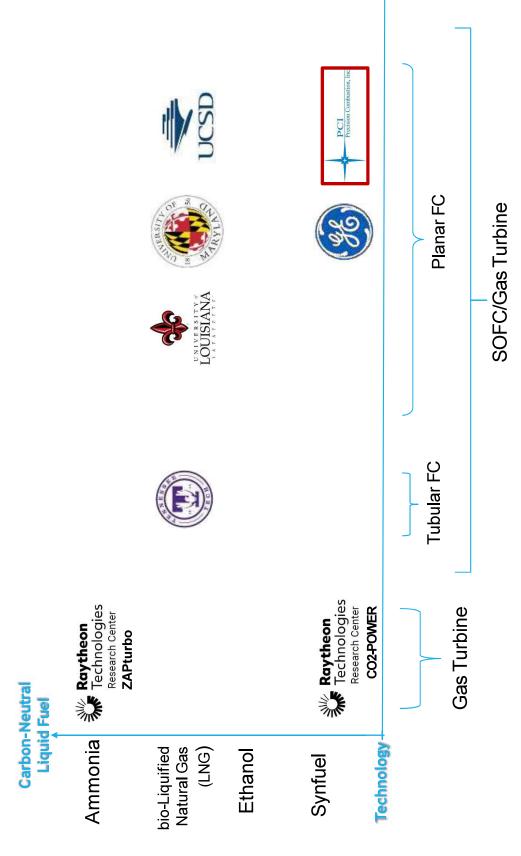


REEACH Program Structure



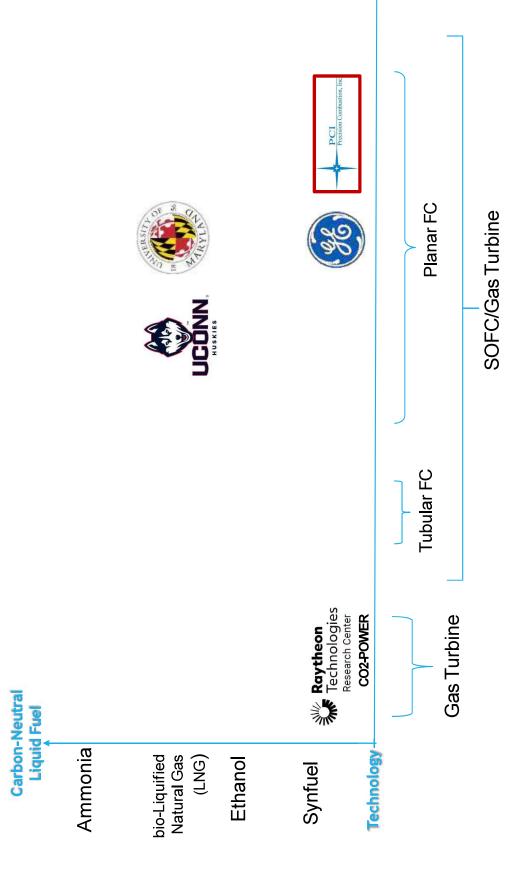


REEACH Phase I Technology Map (01/2021)





REEACH Phase II Technology Map (1/2024)



Technical Risks & Potential Mitigations 11/2023

Risks to ultra-high efficiency at an acceptable specific power (weight)

| # | — | 7 | က | 4 | 5 | ဖ |
|------|-----------------------|---------------------|-------------------|------------|----------------------------------|--------------------------|
| Risk | Specific Power (W/kg) | Power Density (W/L) | Durability (SOFC) | Efficiency | Propulsion System Integration | Energy Storage System |

| | | | | | Catastrophic > 0.9 | |
|------------------------|---------------------|-----------------------|-----------------------|--------------|-----------------------------|-------------------------|
| | — | 2 2 | 4 | | Major $0.5 \rightarrow 0.9$ | ds/yr) |
| | 3 | | | | Moderate 0.3 → 0.5 | Consequences (Quads/yr) |
| | | | 9 | | Minor 0.1 → 0.3 | Conse |
| | | | | | Insignificant < 0.1 | |
| Almost Certain >90% | Likely 50% → 90% | Moderate 30% → 50% | Unlikely 10% → 30% | Rare <10% | | |
| | | Likelihood | | | | |

Technical Risks & Potential Mitigations 08/2025

Risks to ultra-high efficiency at an acceptable specific power (weight)

| # | — | 7 | က | 4 | 5 | 9 |
|------|-----------------------|---------------------|-------------------|------------|----------------------------------|--------------------------|
| Risk | Specific Power (W/kg) | Power Density (W/L) | Durability (SOFC) | Efficiency | Propulsion System Integration | Energy Storage System |

| | | | | | U | |
|------------------------|---------------------|-----------------------|-----------------------|--------------|-----------------------------|-------------------------|
| | | | | | Catastrophic > 0.9 | |
| | | 2 | | 4 | Major 0.5 → 0.9 | ds/yr) |
| | 8 | | | | Moderate 0.3 → 0.5 | Consequences (Quads/yr) |
| | | | | | Minor $0.1 \rightarrow 0.3$ | Conse |
| | | | | 9 | Insignificant < 0.1 | |
| Almost Certain >90% | Likely 50% → 90% | Moderate 30% → 50% | Unlikely 10% → 30% | Rare <10% | | |
| | | Likelihood | ' | | | |

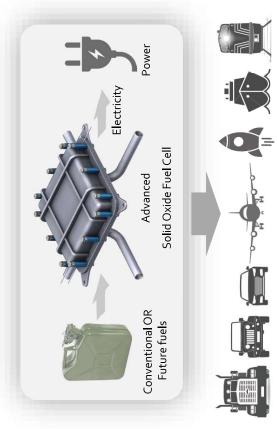
SOFC Performer Details



PI: Subir Roychoudhury **PI**: Subir Roychoudhury **PI**'s email: sroychoudhury@precision-combustion.com

<u>F</u>uel-flexible, <u>Lightweight, Internally-reformed, G</u>as-turbine <u>H</u>ybridized SOFC for <u>Transportation</u> (*FLIGHT*)

| Full Scale Phase II Design Test Article | Solid Oxide Fuel Cell | Synfuel, kerosene, etc. | 1.05 0.5 | TBD TBD | Application 10 Specific (SOFC) | >20% | TBD 100 VDC | <30 min <60 min | 10,000 [TBR] | <1000 |
|---|-----------------------|-------------------------|----------------------------------|---------------------------------|-----------------------------------|--------------------------------|-------------------------------|-----------------|---------------------|------------------------------|
| Unit | | | kW/kg | kW/l | kW | % | Λ | min. | Hrs. | $$/\mathrm{kW}_{\mathrm{e}}$ |
| Parameter | Technology Type | Fuel | Powertrain specific power (peak) | Volumetric power density (peak) | Peak power rating | Fuel to electricity efficiency | Output voltage & type (AC/DC) | Start-up time | Estimated ESPG MTBF | Predicted ESPG CAPEX |

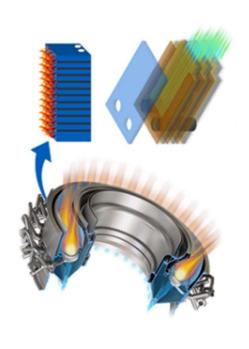


7

PI: Dr. John Hong (Senior Engineer, GE Aerospace) PI's email: John.Hong@GEAerospace.com

WITH GAS TURBINE (GT) GENSET, FOR FUEL-FLEXIBLE & HIGH-EFFICIENCY THRUST AND POWER GENERATION INNOVATIVELY INTEGRATING HIGH POWER DENSITY METAL-SUPPORTED SOLID OXIDE FUEL CELL (MS-SOFC)

| Technology Type S H2, C Fuel | | |
|---|--|----------------------------------|
| | SOFC-GT Hybrid | ybrid |
| | H2, CH4, Kerosene-based fuel (e.g., Jet-A), SAF (e.g., HEFA-SPK) | ene-based , SAF (e.g., PK) |
| Powertrain specific power (peak)* kW/kg > 1 | > 1 | ~1 |
| Volumetric power density (peak)* kW/l TBI | TBD | ~4 |
| Peak power rating* kW > 100 | > 1000 | 2 |
| Fuel to electricity efficiency* 80 | %08 < | %02 < |
| Output voltage & type $(AC/DC)^*$ V TBI | TBD > | > 70V DC |
| Start-up time* min. | Start with GT | LGT |
| Estimated ESPG MTBF* Gas | Gas Turbine Standard | tandard |
| Predicted ESPG CAPEX* \$/kW | ~940 | |

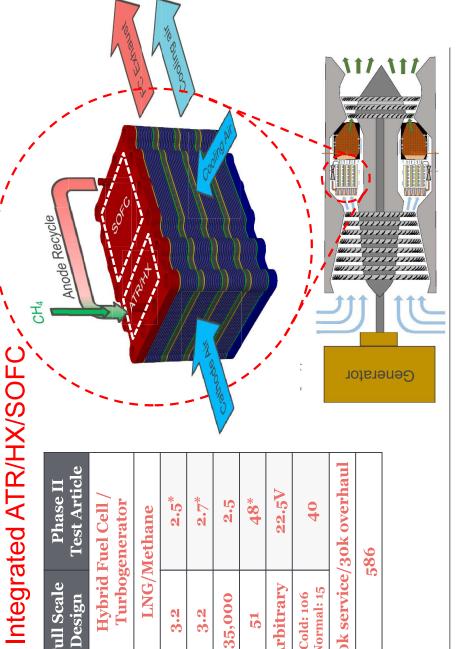


^{*}Current engineering estimates or targets

Hybrid SOFC-Turbogenerator for Aircraft

PI: Christopher Cadou Pl's email: cadou@umd.edu

| | | | |) -)) ? |
|----------------------------------|----------|-------------------------|--------------------------------------|-----------------------|
| Parameter | Unit | Full Scale Design | Phase II Test Article | ``. |
| Technology Type | | Hybrid Turbog | Hybrid Fuel Cell / Turbogenerator | ` |
| Fuel | | I/BNT | LNG/Methane | |
| Powertrain specific power (peak) | kW/kg | 3.2 | *0.0 | |
| Volumetric power density (peak) | kW/1 | 3.2 | * | 1 300 |
| Peak power rating | kW | 35,000 | 2.5 | Bounes |
| Fuel to electricity efficiency | % | 21 | *8* | p p |
| Output voltage & type (AC/DC) | Λ | Arbitrary | 22.5V | ′ |
| Start-up time | min. | Cold: 106 Normal: 15 | 40 | : |
| Estimated ESPG MTBO | Hrs. | 10k service/ | 10k service/30k overhaul | JC |
| Predicted ESPG CAPEX | $$/kW_e$ | rů | 586 | otere |
| | | | | eu: |



High Performance Metal-Supported SOFC System for Range **Extension of Commercial Aviation**

PI: Xiao-Dong Zhou

Pl's email: xiao-dong.zhou@uconn.edu

BREAKTHROUGH EFFICIENCY AND PERFORMANCE IN NATURAL GAS-POWERED METAL-SUPPORTED SOFCS

The University of Connecticut has achieved significant milestones in the development of metal-supported solid oxide fuel cells (MS-SOFCs), demonstrating unprecedented levels of efficiency and performance. Leveraging liquefied natural gas (LNG), propane, and other readily available fuels, these next-generation fuel cells offer enhanced reliability and are well-positioned to serve as a clean, high-efficiency power source for future applications in aviation, data centers, maritime transport, heavy-duty trucking, and beyond.

| Parameter | Unit | Full Scale Design | Phase II Test Article |
|--|---------------|----------------------|--------------------------------------|
| Technology Type | | Next Gene | Next Generation SOFC |
| Fuel | | Natural Gas, Prop | Natural Gas, Propane, Jet Fuels & H2 |
| Powertrain specific power (peak) kW/kg | kW/kg | 1.5 | 1.5 |
| Volumetric power density (peak) | kW/l | 3.5 | 3.5 |
| Peak power rating | kW | 2500 | 5 |
| Fuel to electricity efficiency | % | 80 | 80 |
| Output voltage & type (AC/DC) | Λ | 1,000 | 8V (DC) for 1 kW stack |
| Start-up time | min. | < 30 | < 15 |
| Estimated ESPG MTBF | Hrs. | 12 | 12,000 |
| Predicted ESPG CAPEX | $$/{ m kW_e}$ | \$ | \$125 |



Thank You

Program Director James Seaba **ARPA-E** James.seaba@hq.doe.gov